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TRANSMITTAL LETTER TO THE UNITED STATES
DESIGNATED/ELECTED OFFICE (DO/EO/US)
CONCERNING A FILING UNDER 35 U.S.C. 371

U.S. APPLICATION NO. (If known, see 37 CFR 1.5)

10/069101

INTERNATIONAL APPLICATION NO.

PCT/GB00/03279

INTERNATIONAL FILING DATE

August 24, 2000

PRIORITY DATE CLAIMED

August 24, 1999

TITLE OF INVENTION REINFORCED NODAL STRUCTURE, REINFORCEMENT WITH A CORE OF
EXPANSIBLE MATERIAL AND METHOD OF MOULDING AN ARTICLE

APPLICANT(S) FOR DO/EO/US


Andrew MILLS, Steven COUSINS, Lee BATEUP and Robert BACKHOUSE

Applicant herewith submits to the United States Designated/Elected Office (DO/EO/US) the following items and other information:

1. ☒ This is a **FIRST** submission of items concerning a filing under 35 U.S.C. 371.
2. ☐ This is a **SECOND** or **SUBSEQUENT** submission of items concerning a filing under 35 U.S.C. 371.
3. ☒ This is an express request to begin national examination procedures (35 U.S.C. 371(f)). The submission must include items (5), (6), (9) and (21) indicated below.
4. ☐ The US has been elected by the expiration of 19 months from the priority date (Article 31).
5. ☒ A copy of the International Application as filed (35 U.S.C. 371(c)(2))
 - a. ☒ is attached hereto (required only if not communicated by the International Bureau).
 - b. ☒ has been communicated by the International Bureau.
 - c. ☐ is not required, as the application was filed in the United States Receiving Office (RO/US).
6. ☐ An English language translation of the International Application as filed (35 U.S.C. 371(c)(2)).
 - a. ☐ is attached hereto.
 - b. ☐ has been previously submitted under 35 U.S.C. 154(d)(4).
7. ☐ Amendments to the claims of the International Application under PCT Article 19 (35 U.S.C. 371(c)(3))
 - a. ☐ are attached hereto (required only if not communicated by the International Bureau).
 - b. ☐ have been communicated by the International Bureau.
 - c. ☐ have not been made; however, the time limit for making such amendments has NOT expired.
 - d. ☐ have not been made and will not be made.
8. ☐ An English language translation of the amendments to the claims under PCT Article 19 (35 U.S.C. 371(c)(3)).
9. ☐ An oath or declaration of the inventor(s) (35 U.S.C. 371(c)(4)).
10. ☒ An English language translation of the annexes of the International Preliminary Examination Report under PCT Article 36 (35 U.S.C. 371(c)(5)).

Items 11 to 20 below concern document(s) or information included:

11. ☐ An Information Disclosure Statement under 37 CFR 1.97 and 1.98.
12. ☐ An assignment document for recording. A separate cover sheet in compliance with 37 CFR 3.28 and 3.31 is included.
13. ☒ A **FIRST** preliminary amendment.
14. ☐ A **SECOND** or **SUBSEQUENT** preliminary amendment.
15. ☐ A substitute specification.
16. ☐ A change of power of attorney and/or address letter.
17. ☐ A computer-readable form of the sequence listing in accordance with PCT Rule 13ter.2 and 35 U.S.C. 1.821 - 1.825.
18. ☐ A second copy of the published international application under 35 U.S.C. 154(d)(4).
19. ☐ A second copy of the English language translation of the international application under 35 U.S.C. 154(d)(4).
20. ☒ Other items or information: 1) PCT Request; 2) Intl. Search Report and 3) Intl. Preliminary Examination Report

U.S. APPLICATION NO. (if known, see 37 CFR 1.53) 10/069101		INTERNATIONAL APPLICATION NO. PCT/GB00/03279		ATTORNEY'S DOCKET NUMBER SGU-0050	
21. <input checked="" type="checkbox"/> The following fees are submitted: BASIC NATIONAL FEE (37 CFR 1.492 (a) (1) - (5)): Neither international preliminary examination fee (37 CFR 1.482) nor international search fee (37 CFR 1.445(a)(2)) paid to USPTO and International Search Report not prepared by the EPO or JPO. \$1040.00 International preliminary examination fee (37 CFR 1.482) not paid to USPTO but International Search Report prepared by the EPO or JPO \$890.00 International preliminary examination fee (37 CFR 1.482) not paid to USPTO but international search fee (37 CFR 1.445(a)(2)) paid to USPTO \$740.00 International preliminary examination fee (37 CFR 1.482) paid to USPTO but all claims did not satisfy provisions of PCT Article 33(1)-(4) \$710.00 International preliminary examination fee (37 CFR 1.482) paid to USPTO and all claims satisfied provisions of PCT Article 33(1)-(4) \$100.00 ENTER APPROPRIATE BASIC FEE AMOUNT =				CALCULATIONS PTO USE ONLY <div style="border: 1px solid black; padding: 2px;"> \$ 890.00 </div>	
Surcharge of \$130.00 for furnishing the oath or declaration later than <input type="checkbox"/> 20 <input checked="" type="checkbox"/> 30 months from the earliest claimed priority date (37 CFR 1.492(e)).				<div style="border: 1px solid black; padding: 2px;"> \$ 130.00 </div>	
CLAIMS	NUMBER FILED	NUMBER EXTRA	RATE	\$	
Total claims	16 - 20 =	0	x \$18.00	\$	
Independent claims	3 - 3 =	0	x \$84.00	\$	
MULTIPLE DEPENDENT CLAIM(S) (if applicable)				+ \$280.00	
TOTAL OF ABOVE CALCULATIONS =				\$ 1,020.00	
<input checked="" type="checkbox"/> Applicant claims small entity status. See 37 CFR 1.27. The fees indicated above are reduced by 1/2.				\$ 510.00 +	
SUBTOTAL =				\$ 510.00	
Processing fee of \$130.00 for furnishing the English translation later than <input type="checkbox"/> 20 <input type="checkbox"/> 30 months from the earliest claimed priority date (37 CFR 1.492(f)).				<div style="border: 1px solid black; padding: 2px;"> \$ </div>	
TOTAL NATIONAL FEE =				\$ 510.00	
Fee for recording the enclosed assignment (37 CFR 1.21(h)). The assignment must be accompanied by an appropriate cover sheet (37 CFR 3.28, 3.31). \$40.00 per property +				<div style="border: 1px solid black; padding: 2px;"> \$ </div>	
TOTAL FEES ENCLOSED =				\$ 510.00	
				Amount to be refunded: \$	
				charged: \$	
a. <input checked="" type="checkbox"/> A check in the amount of \$ 510.00 to cover the above fees is enclosed. (check #02180) b. <input type="checkbox"/> Please charge my Deposit Account No. _____ in the amount of \$ _____ to cover the above fees. A duplicate copy of this sheet is enclosed. c. <input checked="" type="checkbox"/> The Commissioner is hereby authorized to charge any additional fees which may be required, or credit any overpayment to Deposit Account No. <u>16-0607</u> . A duplicate copy of this sheet is enclosed. d. <input type="checkbox"/> Fees are to be charged to a credit card. WARNING: Information on this form may become public. Credit card information should not be included on this form. Provide credit card information and authorization on PTO-2038.					
NOTE: Where an appropriate time limit under 37 CFR 1.494 or 1.495 has not been met, a petition to revive (37 CFR 1.137 (a) or (b)) must be filed and granted to restore the application to pending status.					
SEND ALL CORRESPONDENCE TO: Fleshner & Kim, LLP P.O. Box 221200 Chantilly, VA 20153-1200				<div style="text-align: center;">  SIGNATURE Carl R. Wesolowski NAME 40,372 REGISTRATION NUMBER </div>	

Docket No.: SGU-0050PATENT**IN THE UNITED STATES PATENT AND TRADEMARK OFFICE**

In re Application of :

Andrew MILLS, Steven COUSINS, Lee BATEUP :
and Robert BACKHOUSE :Serial No. New U.S. National Phase of :
Intl. Appln. No. PCT/GB00/03279 :

Confirm. No.: Unassigned :

International Filing Date: August 24, 2000 :

U.S. Natl. Phase Filing Date: February 22, 2002 :

For: REINFORCED NODAL STRUCTURE, REINFORCEMENT WITH A
CORE OF EXPANSIBLE MATERIAL AND METHOD OF MOLDING AN
ARTICLE**PRELIMINARY AMENDMENT**Assistant Commissioner for Patents
Washington, D. C. 20231

Sir:

Prior to initial examination on the merits, please amend the above-identified application
as follows:

IN THE SPECIFICATION:

Prior to the first paragraph of the specification at page 1, please insert the following
paragraph:

--This application claims priority to PCT/GB00/03279, filed August 24, 2000,
Publication No. WO 01/14128 A1 in the English language and which claimed priority to GB
Patent Application No. 9920071.9 filed August 24, 1999.--

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IN THE CLAIMS:

A. Please cancel claims 1-16 without prejudice or disclaimer.

B. Please add new claims 17-32 as follows:

17. (New) A method of moulding a reinforced nodal structure which includes laying down a cored reinforcement of constant cross section in and along the channels of a nodal mould and across the nodes thereof by repeated passes along the channels to at least partially fill the channels, closing the mould, and curing resin provided around the reinforcement.

18. (New) A method according to claim 17, wherein the reinforcement is a foam-cored carbon fibre structure.

19. (New) A method according to claim 17, wherein the channels are overfilled whereby closing the mould compresses the reinforcement.

20. (New) A method according to claim 17, wherein the laying down involves relative movement of a feeder head and the mould and control of the feed of reinforcement, all under computer numerical control (CNC).

21. (New) A method according to claim 20, which includes also severing lengths of the reinforcement in the feeder head under CNC.

22. (New) A method according to claim 17, which includes thermally tacking reinforcement to a preceding layer of reinforcement.

23. (New) A method according to claim 17, which includes introducing at least one insert in the mould to divert locally the reinforcement, to provide localised strengthening and/or to provide a mounting point.

24. (New) A machine for laying down reinforcement for a composite moulded nodal frame structure having a feeder head and a mould, means for feeding cored reinforcement of constant cross-section through the feeder head which causes relative movement of the feeder head and the mould so that the reinforcement is laid down in and along a channel of the mould.

25. (New) A machine according to claim 24, wherein the feeder head additionally includes means for severing the reinforcement into lengths.

26. (New) A machine according to claim 24, which is under CNC.

27. A machine according to claim 24, wherein the feeder head includes also a radiant heater.

28. (New) An elongate cored reinforcement of constant cross-section for forming a composite moulded article, the reinforcement comprising an envelope of strength-giving fibres surrounding a core of expansible material.

29. (New) A cored reinforcement according to claim 28, in which the fibres are carbon fibres.

30. (New) A cored reinforcement according to claim 28, in which the envelope of strength-giving fibres has a braided structure.

31. (New) A cored reinforcement according to claim 28, in which the expansible material is a closed cell foam material.

32. (New) A method of moulding a composite article which comprises laying in a mould at least one length of the reinforcement of claim 28, closing the mould, reducing the pressure in the mould to cause expansion of the reinforcement to reduce void space within and around the reinforcement, and curing resin deposited around the reinforcement.

REMARKS

Claims 17-32 are pending. Claims 1-16 are canceled and claims 17-32 are added. Prompt examination and allowance in due course are respectfully solicited.

Respectfully submitted,
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Date: February 22, 2002

WO 01/14128

PCT/GB00/03279

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REINFORCED NODAL STRUCTURE, REINFORCEMENT WITH A CORE OF EXPANSIBLE MATERIAL
AND METHOD OF MOULDING AN ARTICLE

5 The present invention relates to an apparatus for
and method of producing a composite structure and relates
particularly, but not exclusively, to the production of
vehicle structures and the like. The present invention
also relates to a cored reinforcement suitable for
making a composite structure.

10 It is known to produce a composite structure by
laying a plurality of fibres onto a first portion of a
mould having a desired shape and introducing a resin
material either during the laying up process or
thereafter in order to ensure the fibres are bonded to
each other and produce a component of the desired shape.
15 It is also known to employ a mould having inner and outer
portions which, upon assembly, are used to define the
external shape of the component to be moulded.
Additionally, it is also known to employ an injection
technique in which resin is injected or drawn into the
20 mould cavity during the manufacturing process. This
injection step facilitates the impregnation of resin
material between the fibrous structure and fills the
mould cavity thereby to define accurately the final shape
of the desired product.

25 Whilst the above processes provide a perfectly
adequate method of manufacturing a composite component it
does not lend itself readily to the production of
complicated three-dimensional structures. Additionally,
some structures can be somewhat bulkier than might be
30 desired, as producing strong and slim corners and joints
can be problematic.

A particular problem is met in moulding complex,

10069101-041702

multi-noded, frame structures such as vehicle frames. Although it is theoretically possible to use the prior art hand lay-up procedures for these costs would be prohibitive; and, again in the context of vehicle frames, crash-resistance and rigidity in the cage formed by the frame for at least the passenger-carrying part of the vehicle, require great care and control in the formation of the joints at the nodes and in how reinforcement is continued through a node to confer coherence and strength.

Furthermore, use of fibres only as reinforcement will lead to a dense and heavy structure. It is known that the reinforcement can take the form of a carbon-fibre-bound foam core of constant cross-section which is of indefinite length and can therefore be fed onto the mould as desired. This gives a more desirable weight/strength ratio. However, there may still be substantial voids between such reinforcement which become filled with resin, and the desired weight to strength ratio may not be reached.

It is an object of the present invention to provide a machine and process for producing a composite structure which lends itself to the moulding of complicated three-dimensional structures.

Accordingly a first aspect of the present invention provides a machine for laying up moulded resin-based structural composite, wherein the composite is a nodal frame, which includes a feed head for cored reinforcement of constant cross-section, means for controlling the feed of the composite from the head and for severing the composite into fed lengths, and means for causing coordinated relative movement between the head and a

nodal mould whereby to feed lengths of reinforcement into and along the open mould and through the nodes thereof.

The feeding and coordination will be under CNC (computer numerical control).

5 The cross-sectional shape of the cored reinforcement need have no relation to the shape of the mould.

10 The feeding is preferably repeated until the mould is at least filled throughout with the lengths of reinforcement; preferably it is somewhat overfilled so that closure of the mould causes compression of the reinforcement. However, it is not excluded that the mould may be underfilled with the lengths of reinforcement.

15 In a second aspect of the invention a moulding process for making a nodal frame of composite includes providing a length of cored reinforcement of constant cross-section, repeatedly feeding that reinforcement into and along an open nodal mould for the frame through the nodes thereof, severing the length with shorter lengths as necessary to fill the mould, closing the mould, and curing resin provided around the reinforcement.

20 The process may include overfilling the open mould, whereby closing the mould compresses the reinforcement.

25 Again, the cross-sectional shape of the reinforcement need have no relation to the shape of the mould.

30 Another object of the present invention is to reduce the mass of a composite moulded structure or article.

 Accordingly in a third aspect of the invention we use as a reinforcement a cored reinforcement with an

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envelope of strength-giving fibres - most preferably carbon fibres - surrounding a core of expansible (preferably closed cell foam) material. The core is expansible under reduced pressure applied to a closed mould (e.g. to draw resin into the mould) and therefore the finished article contains hardly any or ideally no voids between reinforcements to be occupied by resin; and instead the resin is found exclusively impregnating and encapsulating the strength-giving fibrous structure.

The invention in a fourth aspect therefore provides a method of moulding the composite article which consists of laying in a mould at least one length of reinforcement of constant cross-section (which constant cross-section need have no relation to the shape of the mould), the reinforcement having a core of an expansible (preferably closed cell foam) material, closing the mould, reducing the pressure in the closed mould whereby to cause expansion of the reinforcement to minimise void space within and around the reinforcement, curing resin provided around the reinforcement, and removing the mould. Preferably the strength-giving fibre is carbon fibre. The curing which will be under elevated temperature may have the effect of destroying or partially destroying the foam core.

The result is a cellular structure wherein cured resin encapsulates the reinforcing fibres in such a way as to give a favourable mass to strength ratio because the resin tends to be of higher density than the fibre or, of course, the foam core.

A preferable means and method for laying down the reinforcement in this way, preferably in numerous passes over a nodal mould, are provided by the first and second

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aspects of the present invention.

The strength-giving fibre envelope may be prepared in any form suitable for the purpose. Particularly suitable is a braided structure. For the expansion
5 characteristic of this process, the braiding need not be symmetrical.

In the second or fourth aspect of the invention the resin may be provided by passing it into the mould (e.g. by injection, or by evacuation to draw the resin into the
10 mould) when the mould is closed. Alternatively the resin may be introduced into the mould as the reinforcement is fed into mould. For example, the resin may be fed into the mould as a powder as the reinforcement is fed into the mould. The powder may then be fused (e.g. by a
15 travelling heater) to prevent it from escaping and/or to tack the reinforcement together.

The present invention will now be more particularly described by way of example only with reference to the accompanying drawings in which:

20 Figure 1 is an isometric projection of a vehicle substructure which might be produced in accordance with the method and apparatus of the present invention;

Figure 2 is a diagrammatic representation of a mould structure;

25 Figures 3A and 3B illustrate one possible joint arrangement between segments of the mould structure in top and side view respectively, in an open position;

Figures 4A and 4B illustrate the arrangement of Figures 3A and 3B in closed position;

30 Figure 5 illustrates an encapsulation step with introduction of resin into the interior of the mould structure;

Figure 6 illustrates a node within the composite structure;

Figures 7 and 8 illustrate in side view a machine suitable for laying down reinforcement into a mould structure;

Figure 9 is a front view of a feeder head of the machine;

Figure 10 is a side view of the feeder head of the machine;

Figures 11-18 illustrate various steps involved in the manufacture of a moulded structure in accordance with the present invention;

Figures 19-21 are lateral sections through a mould showing successive stages of loading;

Figure 22 is the same after impregnation and cure; and

Figure 23 is a perspective view and section of a composite prepared according to the process.

Referring now to the drawings in general, but particularly to Figure 1 it will be appreciated that a composite structure such as a vehicle space frame 10 is complex and not easily manufactured. The structure comprises a number of frame members 12 meeting at nodes 13 and may include subassemblies such as crush subsection 14, and components such as a rear strengthening member illustrated generally at 16. Additionally, features such as hinge, bumper or suspension member mounting points may be provided at other positions on the structure.

If the frame structure is to be made integral a collapsible internal mould part (to be described) will have to be used; however if the frame structure is made in two mirror image parts (ie divided along its median

plane) a conventional three-dimensional mould may be used. In the latter case the joining together of the separately made halves by adding further layers over the joint can easily be achieved and be very satisfactory.

5 A simplified form of a collapsible mould structure suitable for creating an integral composite structure of Figure 1 is shown in Figure 2. From Figure 2 it will be appreciated that the mould structure comprises a plurality of linked female mould segments 24 each of which has a channel portion 26 into which the reinforcement is laid during the assembly process described in detail later herein. In an erected position the mould segments 24 act to define the shape of the desired article whilst in a second, collapsed position, 10 they act to allow removal of the moulded article therefrom. In order to facilitate movement between these positions the segments are joined by a reinforced silicone rubber bag 28 which may be inflated by introducing pressurised air or any other suitable fluid through inlet 30, thereby causing the bag to inflate and move the segments 24 to their assembled position. Collapse of the mould segments is achieved by withdrawing the air or other fluid from the bag either by simply releasing it therefrom or by positively withdrawing it. 20 Of course, the shape of the mould here, a simple ovoid, does not correspond with a shape needed for the frame 10; it is given by way of illustration only.

Referring now more particularly to Figures 3 and 4, the segments 24 are jointed together by means of hinge means shown generally at 32. In the particular example 30 the hinge means comprises a flexible web member having a first portion 32a fixedly attached to first segment

10069101-041702

portion 24a and second portion 32b fixedly attached to an adjacent segment portion 24b. The flexible web portion 32 locates each segment 24 relative to its neighbour whilst providing sufficient flexibility for the segments to move between their two positions. In a simplified arrangement the hinge means may comprise a flexible material which forms the pressurising bag 28. Also shown in Figures 3 and 4 are locating means in the form of, for example, tapered pins 34 and associated holes 36 provided on adjacent flange portions 38a, 38b of adjacent mould segments 24a, 24b. Operation of the pressurising means 28 will cause the hinged segments to be moved into interlocking relationship with each other in view of the fact that the pressurising force will be applied in the direction of arrows F of Figures 3 and 4. The force causes the segments to hinge relative to each other into their closed position and ensures the security of the hinged joint.

The segments 24, once erected, act to define a mould structure onto which the composite material may be laid down, as will be described later.

Closure and injection steps are illustrated by reference to Figure 5 from which it will be appreciated that a second part 40 of the mould is brought in to enclose the laid down material 38 and is then secured in position by any suitable means. It will be appreciated that whilst Figure 5 illustrates a total encapsulation type arrangement one need only actually enclose the portions containing the laid down material and, consequently, the mould part 40 need not provide a total enclosure. In order to facilitate the rapid assembly of the tooling one might employ pneumatic or robotic

actuation systems (not shown) which move the second portion or portions 40 of the mould into position and retain them there during the subsequent steps. The act of introducing the second portion 40 preferably causes the reinforcement to be compressed and ensures the fibre structure and any inserts in it are kept still during the subsequent impregnation step. The second portion 40 of the mould is preferably coated with a silicone layer to aid sealing during infusion and release once the component has been cured. By applying a vacuum to the interior of the mould via outlet 41 one can draw resin material from reservoir 42 into the interior of the mould and cause it to pass along the strands of the fibre of the reinforcement via resin inlet 43, thereby passing between and coating the fibre with the resin, which also acts to define the outer surface of the finished article in view of the fact that it contacts the surface of the mould structure itself. The use of a vacuum step is preferred over that of a resin injection step under positive pressure as the vacuum makes the job of sealing significantly easier and reduces loads on the tooling associated with the mould process. Whilst not absolutely necessary, it will be appreciated that a small additionally internal pressure may be applied to the pressurising means 28 to help ensure that the segments 24 seal against the second portion of the mould.

Once the resin is injected, the resin is cured at elevated temperature and the second portion 40 of the mould is removed and the pressurising means deflated for the extraction of the completed space frame such as 10. Any mould/debris is removed from the space frame and the tooling is then cleaned and prepared to re-use once

again. Of course, if the frame is made in halves, the mould channels 26 may be on a permanent, rigid, mould half.

Referring now to Figure 6, it will be appreciated that nodes 13 may be produced by introducing diverting inserts in the form of sections 44 thereby to divert a portion of the reinforcement around the corner created by said section so that the lengths of reinforcement pass from one frame part 12 to another continuously through the node 13.

Figure 7 illustrates in a very simplified form a machine suitable for laying reinforcement onto the mould structure. The machine 60 comprises a support frame 62 having a reinforcement supply head 64, to be described in more detail later, mounted onto a two-axis positioning head 66. The head 66 is mounted on and translatable - arrow X - along bridging member 68 which is, itself, translatable - arrow Y - along frame 62 so as to move in two dimensions. An ovoid mould structure formed of segments 24 is mounted for rotation about a longitudinal axis 63 of the frame 62 such that, on controlled translation of the feed head 64 and rotation of the mould structure it is possible to deposit the reinforcement into the channel 26 of the mould by laying it on and along that channel. This Figure also illustrates the crossover or inter meshing relationship at nodes 13 of the frame thereby to increase the rigidity of the finished frame. A programmable CNC 70 is provided to control the movement of the support frame 62 and feed head 64.

Control of the coordination of the movement of the mould and the feeder head so as to achieve laying-down of

reinforcement in the channels will normally be by the CNC, programmed for the particular frame, ie the particular mould being used: CNC will also determine the rate of feed of reinforcement from the head and if discrete lengths of it are to be placed, for example through a node or at a zone where particularly high stress is expected.

Figure 8 shows a second such machine 60', with a feeder head 64 to which reinforcement comes from a roll 65 carried with the head (in other embodiments, however, the roll may be immobile or independently movable). As before, the head 64 can execute longitudinal or lateral movements over a mould structure, here generally 67 mounted for rotation on axis 63. Under the control of a CNC 70, reinforcement can be laid down either continuously or in discrete lengths along the grooves of the mould structure until these are full or slightly overfull.

Both Figures 7 and 8 have dealt with a fully rotatable mould structure; it is clear however that the mould structure may be stationary, with the feeder head adapted under CNC to move additionally in a Z axis orthogonal to the X and Y arrows, and/or can partially rotate (reciprocate). This will be the case, usually, when the frame to be formed by the machine and process is such as not to require a collapsible mould structure.

Figures 9 and 10 shows the feeder head 64. Reinforcement 38 from the roll 65 or other source is taken by driven feed rollers 45 at a required rate through nozzle 46 with guide wings 47. Reinforcement issuing from the nozzle is pressed into the base of channel 26, or on to a preceding layer of reinforcement,

by pressure roller 28. Cutter 49, which like feed rollers 45 is under CNC control, can operate to sever discrete lengths of reinforcements.

The reinforcement 38 comprises carbon fibre filament 50 surrounding a central core 52 which, in a preferred arrangement, comprises a compressible core such as a foam material. As described later, in respect of Figures 19-23, the reinforcement in a particularly preferred arrangement has an expansible closed cell foam core. It will, however, be appreciated that flexible or non-compressible cores may be used to advantage.

Powdered fusible binder is carried on or in the carbon fibre cover 50. A pulsed infrared heater 53 fuses binder on the surface of any preceding layer of reinforcement to tack the newly applied layer in place under closure of the mould and impregnation, as will be described with reference to Figures 11-18. Whilst the majority of the reinforcement is wound continuously it will be

appreciated that this winding process may be stopped and then recommenced at any position of the mould structure such that localised areas may be provided with additional composite material in order to improve the strength of that portion. In addition to the deposition of discrete lengths of reinforcement it is possible to incorporate

additional fabric, foam and metal inserts into the wound structure as the reinforcement is supplied thereto (such inserts can also, however, be added when winding is complete). Such additions serve to enable the structure to withstand large or localised loads during use and/or provide mounting points for components which must be mounted to the basic mould structure. As showed in Figures 11-18, the mould structure is filled with the

feedstock from feeder head 64 (here shown as delivering a plurality of reinforcements at one pass) whilst at the same time introducing any additional inserts (shown generally at 46) and the mould is then overfilled by a small amount (Figure 14) so that when the mould is closed by second mould part 40' (Figures 15 and 16) the reinforcement is compressed. Resin is then fed in, preferably by vacuum impregnation (Figure 17), to impregnate the voids, and is cured. The mould parts 24, 40' are removed leaving the formed frame member 12.

Figure 19 shows the channel 101 of a mould part 102 being loaded by a feeder head 103 (such as the one described above in respect of Figures 9 and 10) with successive layers of constant cross-section reinforcement 104. The reinforcement here is shown as rectangular cross-section; this is diagrammatic and normally it would be circular or oval in cross-section, with an comparatively soft, expansible, closed cell plastics material foam core enwrapped by an envelope of strength-giving fibres. Particularly suitable for such fibres are carbon fibres and they may be arranged in a braid around the core.

The reinforcement is fed in until the channel is sufficiently full, then as seen in Figure 20 a second part 105 of the mould is placed on and sealed to it. In Figure 21, reduced pressure is applied to the mould to draw resin into it. The resin cannot permeate the closed cell foam core and is instead drawn along the channels formed by the fibre covers of the various lengths of reinforcement. At the same time, the reduced pressure causes an expansion of the foam cores so that the voids between the reinforcements are substantially or even

entirely eliminated, leaving only a comparatively thin network or honeycomb 106 of walls of reinforcing fibre impregnated with resin. This is shown schematically in Figure 22 where the presence of a resin is indicated by thickened honeycomb lines 107. In the process of curing, which is at an elevated temperature, the foam core of the reinforcement may be partially or even completely destroyed or melted but this is of no importance. What is left, as seen in Figure 23, is a honeycomb rigid-walled complete beam structure 108 of very satisfactory strength to weight ratio. Voids where the foam has been destroyed are seen in the darkened areas such as 109.

This construction and type of reinforcement can be used in complex nodal structures, as described above in respect of Figures 1-18, with the reinforcement being taken through the nodes so as to form an integral structure not requiring the making of joints.

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CLAIMS

1. A method of moulding a reinforced nodal structure which includes laying down a cored reinforcement of
5 constant cross section in and along the channels of a nodal mould and across the nodes thereof by repeated passes along the channels to at least partially fill the channels, closing the mould, and curing resin provided around the reinforcement.

10 2. A method according to claim 1 wherein the reinforcement is a foam-cored carbon fibre structure.

15 3. A method according to claim 1 or claim 2 wherein the channels are overfilled whereby closing the mould compresses the reinforcement.

20 4. A method according to any one of the previous claims wherein the laying down involves relative movement of a feeder head and the mould and control of the feed of reinforcement, all under computer numerical control (CNC).

25 5. A method according to claim 4 which includes also severing lengths of the reinforcement in the feeder head under CNC.

30 6. A method according to any one of the preceding claims which includes thermally tacking reinforcement to a preceding layer of reinforcement.

7. A method according to any one of the preceding

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claims which includes introducing at least one insert in the mould to divert locally the reinforcement, to provide localised strengthening and/or to provide a mounting point.

5

8. A machine for laying down reinforcement for a composite moulded nodal frame structure having a feeder head and a mould, means for feeding cored reinforcement of constant cross-section through the feeder head which causes relative movement of the feeder head and the mould so that the reinforcement is laid down in and along a channel of the mould.

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9. A machine according to claim 8 wherein the feeder head additionally includes means for severing the reinforcement into lengths.

10. A machine according to claim 8 or claim 9 which is under CNC.

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11. A machine according to claim 8, claim 9 or claim 10 wherein the feeder head includes also a radiant heater.

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12. An elongate cored reinforcement of constant cross-section for forming a composite moulded article, the reinforcement comprising an envelope of strength-giving fibres surrounding a core of expansible material.

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13. A cored reinforcement according to claim 12, in which the fibres are carbon fibres.

14. A cored reinforcement according to claim 12 or

207140 10169001

claim 13, in which the envelope of strength-giving fibres has a braided structure.

15. A cored reinforcement according to claim 12, claim
5 13 or claim 14, in which the expansible material is a closed cell foam material.

10 16. A method of moulding a composite article which comprises laying in a mould at least one length of the reinforcement of one of claims 12 to 15, closing the mould, reducing the pressure in the mould to cause expansion of the reinforcement to reduce void space within and around the reinforcement, and curing resin deposited around the reinforcement.

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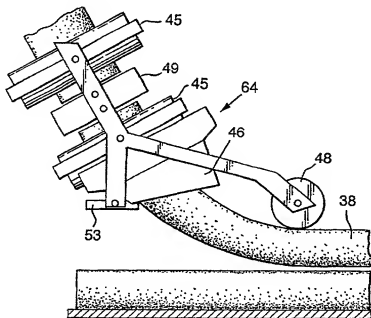
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[Continued on next page]

(54) Title: REINFORCED NODAL STRUCTURE, REINFORCEMENT WITH A CORE OF EXPANSIBLE MATERIAL AND METHOD OF MouldING AN ARTICLE



(57) Abstract: A machine and method for moulding a composite of complex nodal structure includes laying down under CNC a reinforcement (38) of constant cross-section repeatedly into and along channels (26) of a mould (24) and through nodes of the structure, closing the mould, impregnating it with resin and curing the resin.

202140-10169001



WO 01/14128 A1



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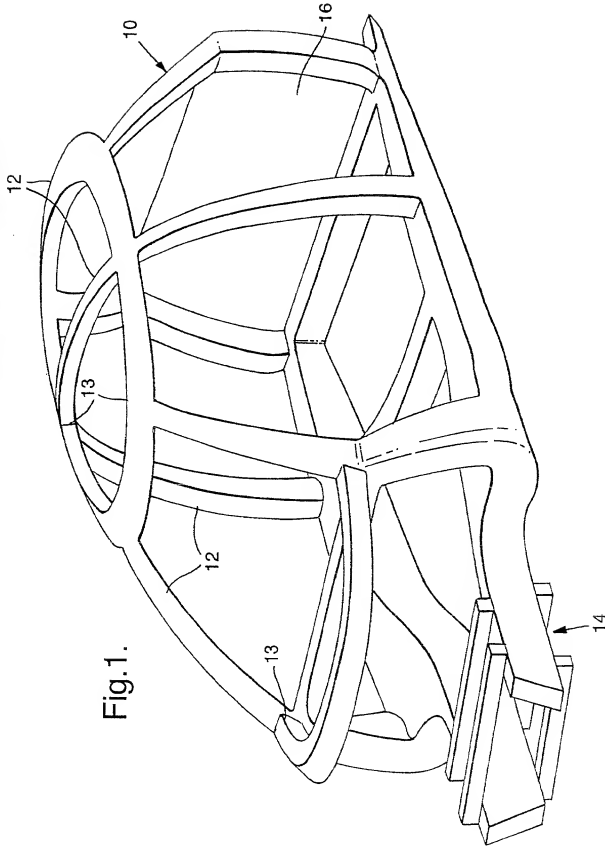


Fig.2.

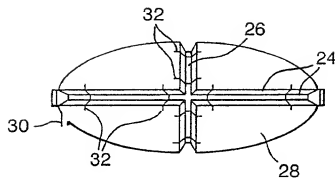


Fig.3A.

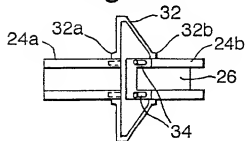


Fig.4A.

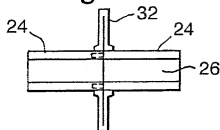


Fig.3B.

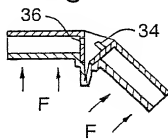


Fig.4B.

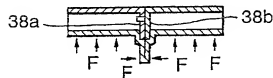


Fig.5.

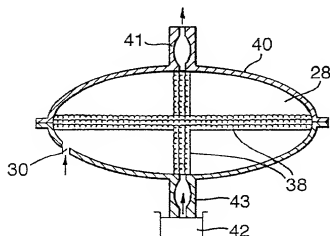


Fig.6.

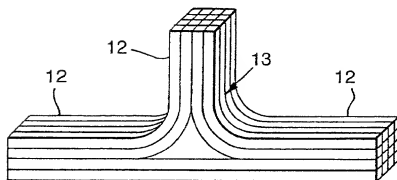
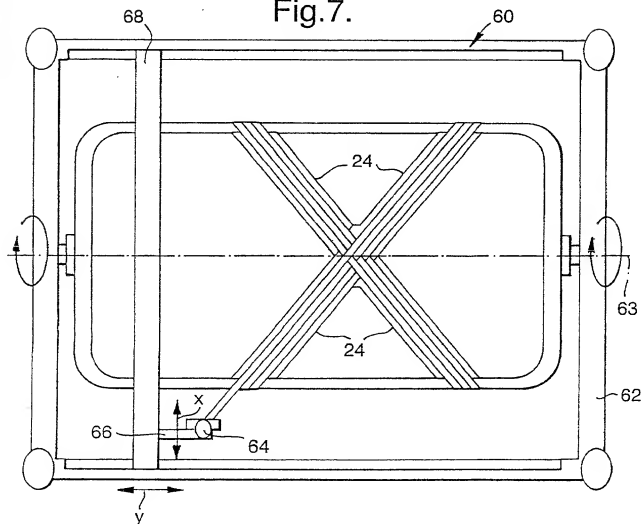
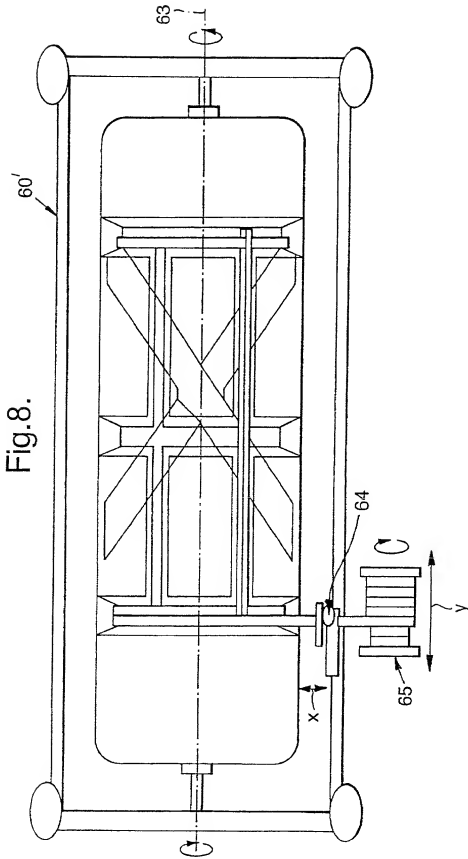


Fig.7.





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Fig.9.

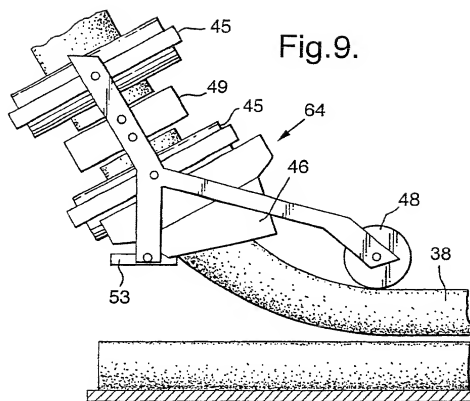


Fig.10.

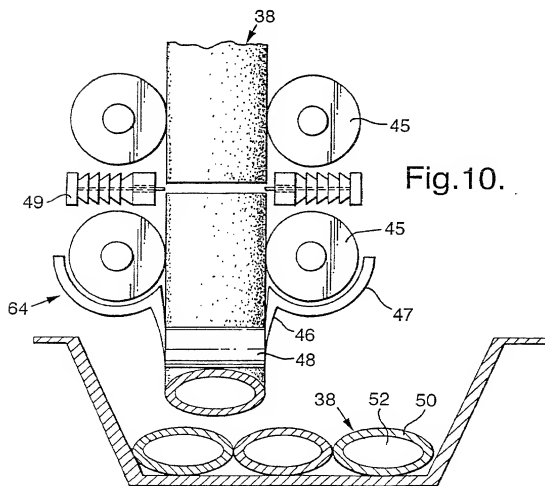


Fig.11.

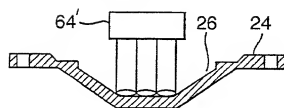


Fig.12.

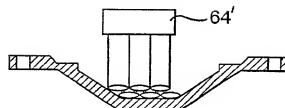


Fig.13.

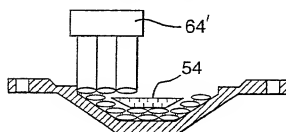


Fig.14.

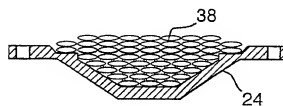


Fig.15.

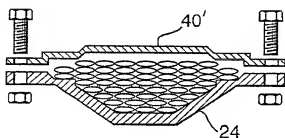


Fig.16.

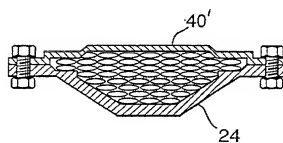


Fig.17.

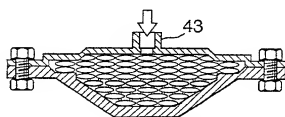
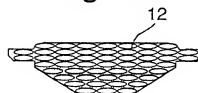


Fig.18.



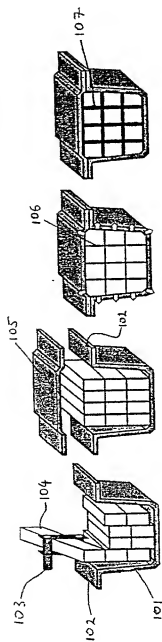


Fig. 22

Fig. 21

Fig. 20

Fig. 19

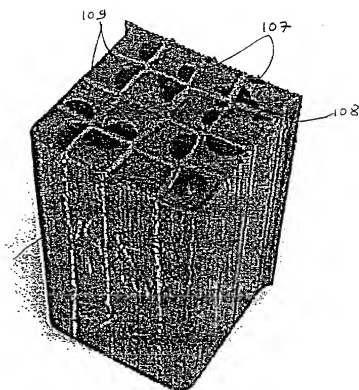


Fig. 23

Docket No.: SGU-0050



DECLARATION AND POWER OF ATTORNEY

As a below named inventor, I hereby declare that:

My residence, post office and citizenship are as stated below next to my name,

I believe I am the original, first and sole inventor (if only one name is listed below) or an original, first and joint inventor (if plural names are listed below) of the subject matter claimed and for which a patent is sought on the invention entitled REINFORCED NODAL STRUCTURE, REINFORCEMENT WITH A CORE OF EXPANSIBLE MATERIAL AND METHOD OF MOLDING AN ARTICLE, the specification of which

☒ [X] is attached hereto ☐ [] was filed on _____ as Application Serial No. _____ and was amended on _____ (if applicable)

I hereby state that I have reviewed and understand the contents of the above identified specification, including the claims, as amended by any amendment referred to above.

I acknowledge the duty to disclose information which is known to me to be material to patentability in accordance with Title 37, Code of Federal Regulations, Section 1.56(a).

I hereby claim foreign priority benefits under 35 U.S.C. 119(a)-(d) or 365 (b) of any foreign application(s) for patent or inventor's certificate, or 365(a) of any PCT international application which designated at least one country other than the United States of America, listed below and have also identified below, by checking the box, any foreign application for patent or inventor's certificate, or of any PCT international application having a filing date before that of the application on which priority is claimed.

Prior Foreign Application(s):

<u>Number</u>	<u>Country</u>	<u>Foreign Filing Date</u> <u>Month/Day/Year</u>
9920071.9	Great Britain	August 24, 1999

I hereby claim the benefit under 35 U.S.C. 119(c) of any United States provisional application(s) listed below.

Application Number(s):

Filing Date (Month/Day/Year)

I hereby claim the benefit under 35 U.S.C. 120 of any United States application(s), or 365(c) of any PCT international application designating the United States of America, listed below and, insofar as the subject matter of each of the claims of this application is not disclosed in the prior United States or PCT international application in the manner provided by the first paragraph of 35 U.S.C. 112, I acknowledge the duty to disclose information which is material to patentability as defined in 37 CFR 1.56 which became available between the filing date of the prior application and the national or PCT international filing date of this application.

**Prior U. S. Application
or PCT Parent Number**

Filing Date (Month/Day/Year)

Parent Patent Number (if applicable)

PCT/GB00/03279	August 24, 2000	
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I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

13
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